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AN INVESTIGATION OF AIRCRAFT HEATERS

XIII - PERFORMANCE OF CORRUGATED AND

NONCORRUGATED FLUTED TYPE EXHAUST

GAS-AIR HEAT EXCHANGERS

By L. M. K. Boelter, A. G. Guibert, M. A. Miller, and E. H. Morrin University of California



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#### NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

#### ADVANCE RESTRICTED REPORT

AN INVESTIGATION OF AIRCRAFT HEATERS

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#### SUMMARY

Thermal and static pressure-drop performance data on three fluted-type heat exchangers are presented. Two of the heaters utilized corrugated surfaces along the fluid passages and the third one was constructed with non-corrugated surfaces. In these tests all heaters were fitted with the same air shroud. Previously reported data taken on the latter heater, but with a different air shroud, are compared with the current data.

Exhaust-gas rates from 3500 pounds per hour to 6600 pounds per hour and ventilating-air rates from 1500 pounds per hour to 4700 pounds per hour were used. Pressure-drop measurements were made across the exhaust-gas and ventilating-air sides of the exchanger under both isothermal and nonisothermal conditions.

The measured thermal outputs and static pressure drops are compared with predicted magnitudes.

#### INTRODUCTION

Two corrugated fluted type heaters (copper and stainless steel) and another heater of the same type but with plain, noncorrugated passages (stainless steel) were tested in the large test stand of the kechanical Engineering Laboratories of the University of California. (See description of this test stand in reference 1.) These heaters are designed for use in the exhaustgas streams of aircraft engines for cabin, wing, and tailsurface heating systems.

The following data were obtained:

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- 1. Weight rates of exhaust-gas and ventilating air through the two sides of the heat exchanger
- 2. Temperatures of ventilating air and exhaust gas at entrance and exit of the heater
- 3. Temporatures of the heater surfaces
- 4. Static pressure—drop measurements on the exhaust—gas and ventilating—air sides of the
  heater under both isothermal and nonisothermal
  flow conditions.

This report is one of a sories of advance restricted reports that describe research being conducted on aircraft heat exchangers at the University of California under the sponsorship and with the financial assistance of the National Advisory Committee for Aeronautics.

DESCRIPTION OF THE HEATERS AND OF

#### THE TESTING PROCEDURE

The fluted-type heaters tested were all prime-surface parallel-flow units. The fluted passages, constructed of either copper or stainless steel, are tapered at each end of the heater.

The corrugated fluted type heaters consist of 34 alternate ventilating—air and exhaust—gas passages in the case of the copper: heater and 52 alternate ventilating—air and exhaust—gas passages in the case of the stainless—steel heater. (See fig. 16.) These passages are formed from sheets of corrugated metal so that the distance between the walls is constant along the passage, that is, the passage is sinuous in character. The corrugations are spaced \$\frac{3}{2}\$ inch apart and are arranged perpendicularly to the direction of flow of the fluids.

The heater, constructed of noncorrugated stainless steel and containing 32 fluid passages, was also tested previously using a different air shroud. The results of the data obtained at that time were reported in reference 1. The air shroud used in the tests reported here differs from the provious one only in the configuration of the inlet air ducts.

Diagrams and photographs of these heaters are shown in figures 1 to 4 and 16.

The weight rates of exhaust gas and ventilating air were obtained by means of callbrated square-edge orifices.

The exhaust-gas temperatures were measured at the inlet and outlet of the heater by means of shielded, traversing thermocouples. Unshielded, traversing thermocouples were used for the ventilating-air temperature measurements.

Temperatures of the heater surfaces were measured at six points by means of thermocouples. (See fig. 16.)
Three of the thermocouples were located on the ventilating—air side of the heater shell, near the exhaust—gas inlet—the other three thermocouples being similarly located near the exhaust—gas outlet. The three thermocouples in each group are spaced at approximately equal intervals around the circumference.

The arithmetic average of the readings of the three thermocouples located near the exhaust-gas inlet is designated as t<sub>1</sub>, whereas the arithmetic average of the readings of the three thermocouples located near the downstream end of the heater is designated as t<sub>2</sub>. (See figs. 1 and 16 and tables I and II.)

Static pressure-drop measurements were made across the ventilating-air and exhaust-gas sides of the heater. Two taps, 180° apart, were installed at each pressure-measuring station.

## MOTATION

- A area of heat transfer, ft2
- Aa total cross-sectional area of the passeges on the ventilating-air side of the heater, ft?

- Ag total cross-sectional area of the passage on the exhaust-gas side of the heater, ft2
- cpa heat capacity of air at constant pressure, Btu/lb oF
- $c_{pg}$  heat capacity of exhaust gas at constant pressure, Btu/1b  $^{o}F$
- D hydraulic diameter, ft
- Da hydraulic diameter on ventilating-air side, ft
- Dg hydraulic diameter on exhaust-gas side, ft
- fc unit thermal convective conductance (av. with length), Btu/hr ft of
- foa unit' thermal convective conductance for the ventilating air (av. with length), Btu/hr ft2 oF
- f unit thermal convective conductance for the exhaust gas (av. with length), Btu/hr ft2 of
- g gravitational force per unit of mass, lb/(lb sec2/ft)
- G weight rate per unit of area, lb/hr ft2
- Ga weight rate per unit of area for ventilating air, lb/hr ft2
- Gg weight rate per unit of area for exhaust gas, lb/hr ft2
- K isothermal pressure drop factor defined by the equation  $\frac{\Delta P}{\gamma} = \kappa \frac{u_m^2}{2g}$
- significant dimension in equations for f along flat plate, ft
- L length of heat transfer surface, ft
- P heat transfer perimeter, ft
- qa measured rate of enthalpy change of wentilating air, Btu/hr or k Btu/hr (= 1000 Btu/hr)\*
- q measured rate of enthalpy change of exhaust gas, Btu/hr or k Btu/hr (= 1000 Btu/hr)\*

<sup>\*</sup> kBtu .designates kilo btu.

- t<sub>1</sub> arithmetic average of three surface temperature measurements taken near the exhaust-gas inlet, OF
- ta arithmetic average of three surface temperature measurements taken near the exhaust—gas outlet, OF
- $T_a$  arithmetic average mixed-mean absolute temperature of ventilating air =  $\frac{T_{a_1} + T_{a_2}}{2} + 460$ , or
- $T_{av}$  arithmetic average mixed-mean absolute temperature of fluid =  $\frac{T_1 + T_2}{2}$ , or
- Tg arithmetic average mixed-mean absolute temperature of exhaust gas =  $\frac{{}^{7}g_{1} + {}^{7}g_{8}}{2} + 460$ , OR
- T<sub>1</sub> mixed-mean absolute temperature of fluid at entrance section (point 1), OR
- Ta mixed-mean absolute temperature of fluid at exit section (point 2), OR
- Tiso mixed-mean absolute temperature of fluid for iso-thermal pressure drop tests, OR
- um mean velocity of fluid at minimum cross-sectional area of fluid passages, ft/sec
- U over-all unit thermal conductance, Btu/hr ft or
- (UA) over-all thermal conductance, Btu/hr F
- W weight rate of fluid, lb/hr
- Wa weight rate of air, lb/hr
- Wg weight rate of exhaust gas, lb/hr
- Y<sub>1</sub> weight density of fluid at entrance to heating section (point 1), lb/ft<sup>3</sup>
- ΔP static pressure drop, lb/ft2
- $\Delta P_a$  total static pressure drop on ventilating—air side, lb/ft<sup>2</sup>

 $\Delta P_{a}^{l}$ , total static pressure drop on ventilating—air side, inches  $H_{a}O$ 

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- $\Delta P_g$  total static pressure drop on exhaust-gas side,  $lb/ft^2$
- ΔP total static pressure drop on exhaust-gas side, inches H<sub>2</sub>O
- ΔP<sub>duct</sub> isothermal static pressure drop along inlet and outlet ducts of the air shroud, lb/ft<sup>2</sup>
- $\Delta P_{htr}$  isothermal static pressure drop along the heater passages only,  $lb/ft^2$
- $\Delta P_{iso}$  total isothermal static pressure drop along heater and ducts at temperature  $T_{iso}$ ,  $lb/ft^2$
- isothermal friction factor defined by the equation,  $\frac{\Delta P}{Y} = \int \frac{L}{D} \frac{u_m^3}{2\varepsilon}$
- Δt<sub>m</sub> logarithmic mean temperature difference defined by equation (4), OF
- $\Delta T_a$  difference between mixed-mean temperature of ventilating air at sections defined by points 1 and 2 =  $(T_{a_2} T_{a_1})$ ,  ${}^{O}F$
- difference between mixed-mean temperatures of exhaust gas at sections defined by points 1 and  $\mathcal{Z} = (\tau_{g_1} \tau_{g_2})$ , of
- μ viscosity of fluid; lb sec/ft2
- mixed-mean temperature of ventilating air at entrance section (point 1), or
- mixed-mean temperature of ventilating air at exit section, (point 2), of
- mixed-mean temperature of exhaust gas at entrance section (point 1), of
- fg2 mixed-mean temperature of exhaust gas at exit section (point 2), of
- Re Reynolds number = GD/3600 µ g

#### METHOD OF ANALYSIS

#### Heat Transfer

The thermal output of the heaters was determined by the enthalpy change of the ventilating air:

$$q_a = W_a c_{p_a} (\tau_{a_a} - \tau_{a_1})$$
 (1)

in which  $c_{p_a}$  was evaluated at the arithmetic average ventilating—air temperature as a good approximation. A plot of  $q_a$  against  $W_a$  at constant values of the exhaust—gas rate  $(W_p)$  is shown in figures 6 and 10.

On the exhaust-ges side of the heater:

$$q_g = V_g c_{p_g} (\tau_{g_1} - \tau_{g_2})$$
 (2)

where  $c_{p_g}$  was evaluated for air at the arithmetic average exhaust-gas temperature.

The over-all thermal conductance UA was evaluated from the expression

$$q_n = (UA) \Delta t_{lm}$$
 (3)

where  $\Delta t_{lm}$  is the log-mean temperature difference defined by the equation

$$\Delta t_{lm} = \frac{(T_{g_1} - T_{a_1}) - (T_{g_2} - T_{a_2})}{\ln \frac{(T_{g_1} - T_{a_1})}{(T_{g_2} - T_{a_2})}}$$
(4)

The variations of UA, with  $W_a$  and  $W_g$  are shown graphically in figures 7 and 11. The termal output of the heater for values of  $\Delta t_{1m}$  other than those used here

may be predicted by determining UA at the actual weight rates from figures 7 and 11 and using these magnitudes in equation (3)\*

Predictions of the magnitudes of the over-all thermal conductance (UA) were attempted. The expression

$$UA = \frac{1}{\left(\frac{1}{f_c A}\right)_{R} + \left(\frac{1}{f_c A}\right)_{R}}$$
 (5)

was used (reference 2, equation (4)).

The heat transfer area (A) is based upon the heat transfer perimeter measured at the center of the fully fluted section of the heater and upon a longth which consists of that of the fully fluted center section plus one half the longth of each of the tapered ends. For example, in the case of the stainless steel noncorrugated heater (see data on fig. 16):

Length of fully fluted section, 0.917 ft

Length of each tapered end, 0.354 ft

Heat transfer perimeter at section A-A, 5.36 ft = P

The equivalent length of heat transfer surface is then

$$L = \frac{0.354}{2} + 0.917 + \frac{0.354}{2} = 1.27 \text{ ft}$$

Heat transfer area,  $A = PL = 5.66 \times 1.27 = 7.19 \text{ ft}^2$ 

The choice of this length is somewhat arbitrary but it probably yields a conservative value of the over-all thermal conductance (UA).

The unit thermal conductances  $f_{ca}$  and  $f_{cg}$  on the ventilating—air and exhaust—gas sides, respectively, are evaluated from the following equations:

<sup>\*</sup>See alternate method for computing heater output for the case when only the initial temperatures of the air and gas are known (reference 12).

(a) For the stainless steel noncorrugated fluted heater

$$f_{c_a} = 5.56 \times 10^{-4} T_a^{0.896} \frac{G_a}{D_a^{0.8}}$$
 (6)

and

$$f_{cg} = 5.56 \times 10^{-4} T_{g}^{0.888} \frac{G_{g}^{0.8}}{D_{g}^{0.8}}$$
 (7)

in which D is the hydraulic diameter. These equations are valid for the calculation of the unit thermal conductance  $(f_c)$  for forced convection in smooth, straight ducts in which the fluid and hoat-flow mechanisms correspond to those in the turbulent regime.\* The values of the thermal resistances  $(1/f_cA)_a$  and  $(1/f_cA)_g$  can also be obtained by use of chart B of references 1 and 2. (See reference 2 for the derivation of equations (6) and (7).)

(b) For the copper and stainless steel corrugated fluted heaters:

$$f_{c_a} = 9.36 \times 10^{-4} \, T_a \, \frac{G_a^{\circ.8}}{1^{\circ.8}}$$
 (8)

and

$$f_{c_g} = 9.36 \times 10^{-4} T_g \frac{g_{c_s}^{0.8}}{10.8}$$
 (9)

These expressions are based on data for hoat transfer by forced convection over flat plates of length 1, measured in the direction of the flow of the fluid. Equations (8) and (9) are valid only in the region downstream from the point where the flow in the retarded layer along the plate has changed from laminar to turbulent flow. The flow in

<sup>\*</sup>See Discussion of this report for comment on effect of diameter to length ratio for ducts or channels.

the retarded layer near the leading edge of the plate is usually laminar, in which case the unit thermal conductance ( $f_c$ ) is a function of the 0.5 power of G and of the -0.5 power of  $I_c$ .

The use of these equations, based on flat-plate hest-transfer data long the corrugated passages of the heaters tested, implies that a retarded layer is initiated at the crest of each corrugation because the value of the significant dimension 1 was taken to be \$\frac{3}{2}\$inch, the distance between successive crests of the corrugated passages.

# Sample Calculation of (UA)

(For run 49 on noncorrugated fluted type heater. Data from table I and fig. 15.)

(a) Computation of 
$$f_{c_a} = 5.56 \times 10^{-4} T_a^{0.896} \times \frac{G_a^{0.8}}{D_a^{0.8}}$$

$$T_{a.} = \frac{98^{\circ} + 252^{\circ}}{2} + 460^{\circ} = 635^{\circ} R$$

$$G_a = \frac{W_a}{A_a} = \frac{4550 \text{ lb/hr}}{0.112 \text{ ft}^2} = 40,600 \text{ lb/hr ft}^2$$

$$D_a = 4 \times \frac{A_a}{\text{wetted perimeter}} = 4 \times \frac{0.112}{7.60} = 0.0589 \text{ ft}$$

$$f_{c_a} = 5.56 \times 10^{-4} (635)^{0.896} \times \frac{(40,600)^{0.8}}{(0.0589)^{0.8}}$$

= 
$$32.0$$
 Btu/hr ft<sup>2</sup> °F

(b) Computation of 
$$f_{cg} = 5.56 \times 10^{-4} \text{ Tg}^{0.88} \times \frac{\text{Gg}^{0.8}}{\text{Dg}^{0.8}}$$

$$T_g = \frac{1420^\circ + 1372^\circ}{2} + 460^\circ = 1850^\circ R$$

$$G_g = \frac{6670 \text{ lb/hr}}{0.194 \text{ ft}^2} = 34,400 \text{ lb/hr ft}^2$$

$$D_g = 4 \times \frac{A_g}{\text{weited perimeter}} = 4 \times \frac{0.194}{7.68} = 0.101 \text{ ft}$$

$$f_{cg} = 5.56 \times 10^{-4} (1850)^{0.885} \times \frac{(34,400)^{0.8}}{(0.101)^{0.8}}$$

= 34.8 Btu/hr ft2 oF

(c) Computation of UA = 
$$\frac{1}{(f_c A)_a} \frac{1}{(f_c A)_g}$$

$$\frac{1}{32.0 \times 7.19} + \frac{1}{34.8 \times 7.19} = \frac{1}{0.00435 + 0.00400}$$

. UA = 120 Btu/hr °F

The value of UA obtained from the laboratory measurements of  $\mathbf{q}_a$  was (see fig. 7):

$$UA = \frac{q_{a}}{\Delta t_{lm}} = \frac{170,000 \text{ Btu/hr}}{1130^{\circ}F} = 150 \text{ Btu/hr}^{\circ}F$$

Percent deviation of the predicted magnitude from the measured magnitude was -20 percent.

Pressure Drop

Isothermal Pressure Drop. Because of the complex curvature of the inlet and outlet ventilating air ducts, the isothermal pressure drop through these ducts could not be predicted satisfactorily. However, due to the fact that these inlot and outlet ducts were used with both the plain and corrugated fluted type heaters, the influence of the corrugations of the heater motal upon the pressure drop along the fluid passages could be determined from the total measured pressure drop along both the heater and air ducts.

The isothermal pressure drops through the corrugatedfluted passages on the air side of the heater wore calculated in the following manner:

(a) The pressure drop along the air passages
of the plain, uncorrugated fluted passages was calculated by means of the

$$\frac{\Delta P_{\text{htr}}}{\gamma} = \zeta_{\text{iso}} \frac{L}{D} \frac{u_{\text{m}}^2}{2g} \qquad (10)$$

where iso was taken as the friction factor for connercial tubes.

(b) This pressure drop was subtracted from the total drop in pressure across the heater (ΔP<sub>iso</sub>) to obtain the losses in the inlet and outlet ducts

$$\Delta P_{\text{duct}} = \Delta P_{\text{iso}} - \Delta P_{\text{htr}}$$
 (11)

(c) Since the same inlet and outlet ducts were also used on the corrugated fluted heaters, the duct losses calculated from equation (11) using the measurements on the non-

corrugated fluted unit were subtracted from the total measured pressure drop across the corrugated fluted heater, yielding approximately\* the pressure drop in the corrugated air passages. Thus equation (11) was used to compute the pressure drop along the air passages of the corrugated fluted heaters.

'(d) The friction factor was then obtained from equation (10).

In this manner the friction factor, \$\int\_{\begin{subarray}{c} 1 \text{ so} \end{subarray}} for the corrugated surfaces was computed and is tabulated in tables III and IV.

In order to utilize existing data for fittings, and so forth, for the estimation of the pressure drop in the air ducts, equation (10) is used in the form

$$\frac{\Delta P}{\gamma} = K \frac{u_m^2}{2g} \tag{12}$$

In the case of the frictional pressure drop in straight ducts.

$$K = \left(\zeta_{iso} \frac{L}{D}\right) \tag{13}$$

By evaluating the magnitude of K in equation (12) for the complex inlet and outlet air ducts, it was found that the pressure drop through those ducts could be calculated approximately by using a value of K=1.0 to 1.5, corresponding to sharp  $90^{\circ}$  bends. (See references 4 to 8.) Hence, the function of these ducts in conducting the ventilating—air through the heater passages is approximately that of a  $90^{\circ}$  bend.

On the exhaust-gas side of the heaters, the isothermal friction factor  $S_{180}$  was calculated by means of equation (10). In this case,  $\Delta P_{180} = \Delta P_{htr}$ , because the pressure drop along the ducts leading to and from the heater was negligible compared to that along the fluted exhaust-gas passages.

<sup>\*</sup>It should be stated that the pressure drop along the air outlet duct may not be the same when used in the corrugated and noncorrugated units because the heater passages affect the flow conditions at the entrance to the outlet air duct.

Nonisothermal pressure drop.— The nonisothermal static pressure drop of either fluid through the heat exchanger was predicted from isothermal measurements by means of equation (6) of reference 1.

$$\Delta P = \Delta P_{iso} \left( \frac{T_{av}}{T_{iso}} \right)^{1.15} + \left( \frac{G}{3600} \right)^{s} \frac{1}{Y_{ig}} \left( \frac{T_{z}}{T_{i}} - 1 \right) \quad (14)$$

in which  $\Delta P_{180}$  is the total measured isothermal static pressure drop (due to friction alone) at temperature  $T_{180}$ ,  $T_{1}$  and  $T_{2}$  are the mixed-mean absolute temperatures of the fluid at the inlet and outlet of the heater, respectively,  $T_{20}$  is the arithmetic average of  $T_{1}$  and  $T_{20}$ . G is the fluid rate per unit cross-sectional area and  $Y_{1}$  is the weight density, evaluated at temperature  $T_{1}$  of the fluid at the inlet to the heater.

A comparison of measured and predicted nonisothermal pressure drops across each side of the heater is presented in tables V and VI and is shown graphically in figures 8, 9, 12, 13, and 14.

Heat transfer and pressure-drop data for the three heat exchangers are presented in tables I and II.

#### DISCUSSION

The results of the tests on the three fluted-type heat exchangers are shown graphically. The results obtained for the plain-fluted heater are given in figures 6 to 9 and those for the two corregated fluted heaters (stainless steel and copper) are given in figures 10 to 14.

The corresponding physical dimensions of all three heaters are approximately equal. However, the depth of the ventilating—air passages was slightly less for the corrugated fluted heaters. Also, the stainless steel corrugated fluted heater consisted of sixteen air passages which is the same as the plain fluted heater; whereas, the other corrugated fluted heater, constructed of copper, contained seventeen air passages.

# Comparison of Results on Noncorrugated Heater

# Using Two Different Air Shrouds

The same air shroud was used in all of the tests reported here. A comparison can be made, however, between the present results in the case of the plain fluted heater and those reported for the same heater in reference 1. the latter experiments a longer air shroud was used which afforded the same cross-sectional area for the flow of ventilating air but had longer inlet and outlet ducts, thus probably yielding a more even distribution of air flow (although an analysis of the measurements of surface temp. near the ends of the heater did not indicate any large difference in the performance in this respect for the two air shrouds). Because the flow areas were equivalent, the rates of heat transfer were about the same for the heater using either shroud (compare fig. 6 of this report to fig. 21 of reference 1). However, the isothermal pressure drop was almost doubled in the case of the air shroud with the shorter inlet and outlet ducts. All of this increase in pressure drop can be ascribed to the greater curvature of the ducts, in which the air is turned in order to flow through the air passages.

Thus, it may be said that if the longer air shroud were also used on the corrugated fluted heaters, the rates of heat transfer would be about the same as for the shorter shroud but the pressure drops would be decreased appreciably. It is undoubtedly true that in many actual installations a limited space would not permit the use of the longer (lower-pressure drop) air shroud.

#### Heat Transfer

A comparison of the results on the three heaters using the shorter (higher-pressure drop) air shroud reveals that the use of corrugated fluted passages yields thermal conductances approximately 45 percent greater than those obtained with the plain, noncorrugated passages. (Compare figs. 7 and 11.) The rates of heat transfer for the two corrugated fluted heaters were about equal, although the copper heater heat-transfer area was slightly greater due to the additional air passage.

l. Moncorrugated fluted heater. The predicted overall thermal conductance for the plain fluted heater was 20 to 25 percent lower than the value based upon computations of laboratory measurements. (See fig. 7.) Part of this discrepancy is due to the inability to predict the mechanisms of heat transfer along the tapered ends of the fluted passages. A heat-transfer mechanism equivalent to that along the fully fluted center section of the heater was used along the tapered ends in the prediction calculations. Equations (6) and (7), used in the prediction of (UA) for the plain fluted heater, are based on data taken in smooth ducts where the hydraulic diameter of the passage is used as the significant dimension D. The use of the multiplier (1 + 1.1 D/L) in equations (6) and (7) to account for the higher unit thermal conductance near the entrance of a tube or channel would have yielded magnitudes of (UA) about 7 percent higher than those which were obtained without employing this correction. \* (See appendix of roference 9 for a discussion of this correction.)

Corrugated fluted heaters .- The predicted overall thermal conductances (UA) for the corrugated fluted heaters agree well (within 10 percent) with the values derived from laboratory measurements. In these calculations the unit thermal conductance (fc) on either side of the heater was calculated by means of equations (8) and (9), which are based on heat-transfer data from smooth flat plates. The value of the significant dimension was taken to be the distance between successive crests of the corrugations (i.e., the wave length, in.). This choice is equivalent to stating that a retarded layer is initiated at the crest of each corrugation along the walls of the fluid passages and, therefore, the mechanism of heat transfor is the same as that along successive idealized flat The use of the multiplier (1 + 1.1 D/L) is not nocessary for equations (8) and (9), which evaluate the average unit thermal conductance for the length 1. It should also be stated that equations (8) and (9) should be used in regions of systems in which the retarded layer has changed from laminar to turbulent flow. Thus the equations are more applicable for long, flat plates, over which the laminar retarded layer adjacent to the leading edge does not extend along an approciable portion of the flat plate. The curvature of the corrugated passages in this case probably maintains a turbulent retarded layer rather than a laminar layer.

<sup>\*</sup>Laboratory experiments are now being conducted in order to determine the validity of this correction factor.

\*\*Equations (8) and (9) are used without a correction factor for the D/L effect because the unit thermal conductance is expressed as a function of 1.

## .Isothermal Static Pressure Drop

1. Pressure drops through heater. - The pressure drop along the air side of the corrugated fluted heater was 65 percent greater then that along the air side of the noncorrugated heater. On the exhaust-gas side the corresponding increase was about 130 percent. The fractional increase in pressure drop on the air side was less than that on the gas side because the losses in the air inlet and outlet ducts were about one-half the total loss and were the same on both the corrugated and noncorrugated fluted heaters.

The agraement between the isothermal friction factor iso for the exhaust-gas side of the heater, based on laboratory measurements, and that taken for commercial tubing is excellent. (See fig. 15.) This indicates that the method described above for the calculation of the static pressure drop through the inlet and outlet fir ducts; namely, substracting the over-all static pressure drop along the air massages (computed from commercial tube friction factor data) from the total pressure drop across the ducts and the heater, is probably adequate.

If the magnitude of the Reynolds number were the same on both sides of the two corrupated fluted heaters, the magnitude of the friction factor for the corrugated passages would be bout the some. An iran-ction of tables III and IV and figure 15 reveals this to be approximately true. The fact that the gas and air nessages are not alike in share, that the reseages are tenered at each and, and that all of the wetted marinater on the air side does not contain corrugations could account for the differences found. A calculation of the isotherm-1-friction factor from the data of Norris and Enofford (reference 10) for similarly shaped corrugated surfaces revealed a value of  $\zeta_{180} = 0.109$  at a magnitude of Reynolds number of 12,400. At the lowest weight rate used in the tests described in this report (Reynolds number = 10,000) the friction factor was equal to 0.118 for the corner corrugated fluted heater and 0.137 for the stainlass-steel heater. The Reynolds number was evaluated by using the hydraulic diameter for the significant dimension D. A similar value of  $\zeta_{\rm iso}$ was obtained for the sinuous passages on the air side of a cross-flow-type heater described in reference 11.

2. Pressure drop through air ducts.— The complex shape of the inlet and outlet air ducts does not permit a simple prediction of the isothermal pressure drop through these units. In the expression

$$\frac{\Delta P}{\gamma} = K \frac{u_{\dot{m}}^2}{2g} \tag{12}$$

a value of K equal to about 1.0 to 1.5 was found to be appropriate in calculating the duct losses. This magnitude of K is equivalent to that used in computing the pressure drop along a 90° elbow, which the ducts resemble.

# Nonisothermal Static-Pressure Drop

The prediction of the nonisothermal static-pressure drop from the measured isothermal pressure drop, by means of equation (14), was successful in all but one instance. (See figs. 8, 9, 12, 13, and 14.)

The slopes of the nonisothermal pressure drop against weight-rate curves are greater than the isothermal curves when the fluid is cooled (gas side) and less when the fluid is heated (air side). An inspection of equation (14) reveals the basis for those facts.

## . Heater Surface Temperatures

The heater surface temperatures (see tables I and II) appear to be lower for the runs on the stainless—steel heaters than for those on the copper heater. The difficulties encountered in obtaining temperature measurements by the use of thermocouples might account for this deviation.

#### CONCLUSIONS

1. The thermal performance of the noncorrugated fluted heater can be predicted to within 20 to 25 percent by means of equations (5), (6), and (7), based on heat-transfer data of smooth ducts.

- 2. The thermal performance of the corrugated fluted heaters can be predicted to within 5 to 10 percent by means of equations (5), (8), and (9), based on heat-transfer data of flat plates, using the wave lenth of the corrugations as the equivalent flat-plate length.
  - 3. The nonisothermal pressure drop of all the heaters can be adequately predicted from the isothermal pressure drop by means of equation (14).
- 4. The heaters constructed with corrugated passages instead of plain passages yielded about 45 percent higher rates of heat transfer but increased the pressure drop by about 65 percent on the air side and by about 130 percent on the exhaust—gas side of the heater.
  - 5. The isothermal friction factor along the corrugated passages was two to three times that for the plain passages.
  - 6. The inlot and outlet air ducts accounted for about 83 percent of the total measured is othernal pressure drop in the case of the noncorrugated heater and about 50 percent in the case of the corrugated heater.
  - 7. The heat-transfer rates and pressure drops were approximately equal for the two corrugated fluted heaters (copper and stainless steel).
  - 8. The use of an air shroud with abrupt inlet and outlet ducts on the plain fluted heater yielded the same rates of heat transfer but 100 percent greater pressure drops than did the use of a shroud with longer inlet and outlet ducts.

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#### REFERENCES

- 1. Boelter, L. M. K., (Miller, M.) A., Sharp, W. H., Morrin, E. H., Iversen, H. W. and Mason, W. E.: An Investigation of Aircraft Heaters. IX Measured and Prodicted Performance of Two Exhaust Gas—Air Heat Exchangers and an Apparatus for Evaluating Exhaust Gas—Air Exchangers. NACA A.R.R., March 1943.
- 2. Martinelli, R. C., Weinberg, E. B., Morrin, E. H., and
  Boelter, L. M. K.: An Investigation of Aircraft
  Heaters. III Measured and Predicted Performance of
  Double Tube Heat Exchangers. NACA A.R.R., October 1942.
- 3. Martinelli, R. C., Weinberg, E. B., Morrin, E. H., and Boelter, L. M. K.: An Investigation of Aircraft Heaters. IV Moasured and Predicted Performance of Longitudinally Finned Tubes. NACA A.R.R., October 1942.
- 4. O'Brien, Morrough P., and Hickox, George, H.: Applied Mechanics. McGraw-Hill Book Co., New York, N.Y., 1937, pp. 211, 350.
- 5. Committee A-9 Aircraft Air Conditioning Equipment: Airplane Heating and Ventilating Equipment. Engineering Data. SAE Jour., January 1943.
- 6. Daugherty, R. L.: Hydraulics. A fest on Practical : Fluid Mechanics. McGraw-Hill Book Co., New York, N. Y., 1937, p. 326.
- 7. Gibson, A. H.: Hydraulics and its Application. D. Van Nostrand Co., New York, N. Y., 1925, pp. 251-255.
- 8. Dodge, Russell A., and Thompson, Milton J.: Fluid Mechanics. McGraw-Hill Book Co., New York, N. Y., 1937, pp. 215-216.
- 9. Boelter, L. M. K., Dennison, H. G., Guibert, A. G., and Morrin, E. H.: An Investigation of Aircraft Heaters. X Measured and Predicted Performance of a Fluted—Type Exhaust Gas and Air Heat Exchanger. NACA A.R.R., March 1943.
- 10. Norris, R. H., and Spofford, W. A. Eigh-Performance Fins for Heat Transfer. Trans. A.S.M.E., vol. 64, no. 5, July 1942, pp. 489-496.

- 11. Boelter, L. M. K., Dennison, H. G., Guibert, A. G., and Horrin, E. H.: An Investigation of Aircraft Heaters. XII Performance of a Formed-Plate Crossflow Exhaust-Gas and Air Heat Exchanger. NAOA A.R.R., May 1943.
- 12. Martinelli, R. C., Morrin, E. H., and Boelter,
  L. M. K.: An Investigation of Aircraft Heaters.
  VI Heat Transfer Equations for the Single
  Pass Longitudinal Exchanger. MACA A.R.R.,
  December 1942.

TABLE I.- EXPERIMENTAL RESULTS ON NON CORRUGATED TYPE HEATER

Run	-	AIR-SIDE						EXHAUST-GAS SIDE						HEATER TEMPS.		OVERALL PERFORMANCE	
	Ta,	Ta,	ΔΤα	Wa	ΔP <sub>a</sub>	g <sub>a</sub> K Btu	Tg,	Îg,	Δĵg	Wg Ib	ΔPg	g, K.Btu	89	t,	t.	Δt	(UA)
No.	°F	*F	*F	M. IP	H <sub>2</sub> O	hr	<b>°</b> F	T	*F	hr	H <sub>2</sub> O	hr		<b>°</b> F	<b>°</b> F	*F	hr <sup>e</sup> T
49	98	252	154	4550	20.3	170	1420	1372	48	6670	4.17	88.1	0.52	528	546	1130	150
50	100	297	197	2970	9.88	142	1407	/368	39	6630	4.65	71.1	0.50	644	659	1190	120
5/	100	376	276	1610	3.36	108	1428	1390	38	6610	4.52	69.1	0.64	791	834	1170	92.5
54	100	236	136	4700	20.1	155	1415	1355	60	4840	2.54	80.1	0.52	466	504	1220	127
53	102	285	181	3000	9.70	/32	1438	1372	66	4840	2.50	87.8	0.67	586	609	1210	109
52	100	353	253	1610	3.3/	98.6	1407	/372	35	4850	2.72	46.7	0.47	714	769	:170	84.5
55	99	220	121	4650	20.0	/36	1424	1342	82	3590	1.32	81.0	0.60	417	441	/230	111
56	103	264	161	2975	9.30	116	1415	1355	60	3560	1.42	58.8	0.51	526	545	1205	96.5
57	103	33/	228	1620	3.30	89.5	1424	1364	60	3560	1.43	58.8	0.66	667	667	1170	76.5

I. COPPE	R CONS	TRUCTION
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	-		_AIR-	·SIDE —			-	E	XHAUS	ST-GAS	SIDE -			HEATER TEMPS.		OVERALL PERFORMANCE	
Run	Ta,	70,	Δῖͼ	Wa	ΔΡά	q,	Îg,	↑ <sub>92</sub>	Δîg	Wg	ΔP	g.	99	t,	t <sub>2</sub>	Δtz	(uA)
No.	°F	<b>•</b> F	<b>*</b> F	<u>lb</u> hr	Inches H <sub>z</sub> O	K. Btu hr	<b>•</b> F	۴F	°F	lb hr	Inches H <sub>2</sub> O	K.Btu hr	•	<b>•</b> F	°F	*F	Btu hr°F
9	104	359	255	3700	22.7	228	1424	1321	100	6620	12./	182	0.80	7/6	704	//30	202
10	106	426	320	2500	12.1	194	1442	1334	108	6560	11.9	194	1.00	816	819	1110	175
//	///	519	408	1550	5.72	153	1442	1377	65	6520	12.3	90	0.65	929	946	1090	140
14	110	322	2/2	3800	22.0	195	/429	/300	129	4886	7.20	174	0.89	628	623	1140	171
/3	112	383	271	2580	11.5	170	1403	1295	108	4886	6.88	145	0.85	7/7	720	1090	156
12	110	467	357	1550	5.35	134	/398	13/2	86	4886	7.00	116	0.86	835	849	1060	127
15	109	291	182	3850	21.9	170	1411	1265	146	3568	4.15	/43	0.84	558	543	1130	150
16	///	357	246	2530	11.4	151	1420	1282	138	3578	4.00	/36	0.90	558	662	1110	136
17	112	444	332	1550	5.05	125	1433	1321	112	3568	3.83	110	0.88	786	798	1090	115

26	100	36/	26/	3400	20.4	215	1414	13/6	98	6497	11.8	175	0.81	472	486	1125	191
27	97	399	302	2490	13.0	182	1398	1321	7 <b>7</b>	6535	11.9	138	0.76	538	555	1100	165
2 <b>8</b>	99	483	384	1550	5.79	144	1420	1351	69	6446	12.1	122	0.86	671	688	1080	/33
29	101	307	206	3470	20.7	174	1416	1278	138	3547	3.80	135	0.78	375	391	1145	151
30	100	356	256	2485	12.0	154	1424	1291	/33	3546	3.70	130	0.84	447	462	1115	138
<i>3</i> /	104	426	322	1590	5.92	124	1420	/3/2	108	3496	3.89	194	0.84	555	573	1090	114

TABLE III

NONCORRUGATED FLUTED TYPE HEATER

Isothermal static pressure drop data

W (1b/hr)	G (lb/hr ft <sup>2</sup> )	ΔP <sub>iso</sub> (lb/ft <sup>2</sup> ) (a)	$= \Delta P_{\text{htr}}$ $(1b/ft^2)$ $(c)$	+ \Delta P_ducts (lb/ft2)	(b)	ζ <sub>iso</sub> <u>L</u> (b)	iso (calcu- lated from data)	Re
	J	<u>.                                    </u>	Ai	r side	<u> </u>	)	<u> </u>	
1500 2500 4000	13,400 22,300 35,700	10.9 28.5 65.0	1.90 4.94 11.4	9.0 23.6 53.6	0.033 .030 .028	0.710 .646 .604	and that they had they	17,100 28,600 45,500
			Ga	s side				
4000 6000 8000	20,600 30,900 41,200	2.23 4.73 8.10	Bellengton and the Bellengton an		0.028 .026 .024	0•329 •323 •298	0.027 .026 .025	45,200 67,800 90,100

<sup>&</sup>lt;sup>a</sup>Pressure drops obtained from plots of  $\Delta P$  against W.

<sup>b</sup> $\zeta_{iso}$  obtained from fig. 7 of reference 3 (friction factor against Reynolds)

 $^{c}\Delta P_{
m htr}$  for air side obtained from predicted ( $^{\zeta}_{
m iso}$  L/D) (See pg. 12)

$$\frac{\Delta P_{\text{htr}}}{\gamma} = \zeta_{\text{iso}} \frac{L}{D} \frac{u_{\text{m}}^2}{2g}$$
 (11)

TABLE IV

CORRUGATED FLUTED TYPE HEATERS

[Nonisothermal static pressure drop data]

	<b></b>					,						
W	G	ΔP <sub>iso</sub>	= $\Delta P_{ m duct}$	+ AP <sub>htr</sub>	ζ <sub>iso</sub> (	iso D	Re					
(1b/hr)	(1b/hr ft <sup>2</sup> )	(1b/ft <sup>2</sup> )	(1b/ft <sup>2</sup> )	(1b/ft²)	(calcu							
		(a)	( <sub>b</sub> )	·	110m							
1. Copper Construction												
			Air si	de								
1500 2500 4000	14,600 24,300 39,800	17.6 43.1 97.7	9.00 23.1 56.1	8.60 20.0 41.6	0.118 .099 .077	2.62 2.20 1.71	17,900 29,700 48,700					
Gas side												
4000 6000 8000	6000 32,000			0-0 had not you Pil.  0-0 and not not be?  0-0 and not not be?	0.111 .099 .090	0.989 •883 •802	65,200 97,600 130,000					
2. Stat	inless-steel	construc	tion			•						
			Air si	de		٠.						
1500 2500 4000	14,400 24,000 39,400	18.2 46.2 109.	9.00 23.1 56.1	9.20 23.1 52.9	0.137 .124 .111	2.88 2.61 2.34	18,600 31,000 50,900					
			Gas si	de		,1						
4000 8000	21,400 32,000 42,800	6.49 12.0 18.6	Old upo (see per und	ويول مدن ويدن ليمن المنا عدن مدن ويدن ليمن عدن مدن ويدن	0.0941 .0778 .0674	0.923 .762 .661	69,000 90,400 138,000					

aPressure drops obtained from plots of ΔP against W.

bΔPduct obtained from data on noncorrugated fluted heater (See text, pg. 12.)

TABLE V

NONCORRUGATED FLUTED TYPE HEATER

[Isothermal static pressure drop data]

Run	<b>W</b> (1b/hr)	G (lb/hr ft²)	Measured isothermal pressure drop $\Delta P_{iso}   \Delta P_{i$		Predicted noniso- thermal pressure drop $\Delta P \mid \Delta P'$ lb/ft <sup>2</sup> in. H <sub>2</sub> 0		Measured noniso- thermal pressure drop  AP 1b/ft² in. H <sub>2</sub> 0		T <sub>1</sub>	T <sub>2</sub>	T <sub>av</sub>
									l		<del></del>
				Exhaust	gas sid	le					
56 53 51	3560 4840 6610	18,000 24,900 34,100	1.81 3.16 5.66	0.35 :61 1.09	5.84 10.1 19.7	1.12 1.95 3.80	7.38 13.0 23.4	1.42 2.50 4.52	1898	1815 1832 1850	186
				Air	side						
52 53 55	1610 3000 4650	14,400 26,800 41,500	12.5 38.9 86.9	2.40 7.50 16.7	19.1 54.0 112.	3.68 10.4 21.6	17.2 50.4 104.0	3.31 9.70 20.0	563 5 <b>62</b> 559	724 7 <b>43</b> 680	64 65 62

<sup>a</sup>These entries are taken from plot of  $\Delta P_g$  against  $W_g$  or  $\Delta P_a$  against  $W_a$  since actual isothermal measurements were at slightly different fluid rates.

$$\Delta P = \Delta P_{iso} \left(\frac{T_{av}}{T_{iso}}\right)^{1.13} + \left(\frac{G}{3600}\right)^{2} \frac{1}{v_{1g}} \left(\frac{T_{2}}{T_{1}} - 1\right)$$

$$\Delta P' = \Delta P \times \frac{12}{62.3} \quad in. \quad H_{2}O$$
(14)

TABLE VI
CORRUGATED FLUTED HEATER

# [Nonisothermal static pressure drop data]

Run	W	G	isoth pres dr	Measured isothermal pressure drop		sted so- sure sp	neasu noni ther press dro	Tı	T2	Tav			
	(lb/hr)	(lb/hr ft <sup>2</sup> )		ΔP'Tiso in. H <sub>2</sub> O	ΔP lb/ft <sup>2</sup>	ΔP <sup>1</sup> in. H <sub>2</sub> O	ΔP lb/ft <sup>2</sup>	ΔP' in. H <sub>2</sub> O	(°R)	(°R)	(°R)		
1.	Copper 1	heater	neater										
	Exhaust gas side												
16 12 9	3580 4890 6620	19,100 26,100 35,400	6.00 10.2 16.6	1.16 1.96 3.20	21.9 34.4 55.7	4.23 6.64 10.8	20.8 36.3 62.8	4.00 7.00 12.1		1837 1794 1781			
~	Air side												
12 16 14	1550 2530 3800	15,100 24,500 36,900	18.5 44.0 90.0	3.57 8.50 17.4	29•9 63•1 125•7	5.76 12.2 24.2	27.8 58.9 114.0	5.35 11.4 22.0	570 571 570	927 817 782	749 694 676		
2.	Stainle	ss steel hea	ter										
	<del></del>	·		Exhau	st gas s	side							
31 27	3500 6540	18,700 34,900	5.10 13.5	0.98 2.60	17.5 46.3	3.38 9.00	20.2 61.7	3.89 11.9		1772 1781	1826 1820		
	Air side												
28 2 <b>7</b> 26	1550 2490 3400	14,900 24,000 32,700	19.5 45.0 78.0	3.76 8.68 15.0	32.1 68.9 115.8	6.20 13.3 22.3	30.0 67.2 105.6	5.79 13.0 20.4	559 557 560	943 859 821	751 708 690		

<sup>a</sup>The entries are taken from plot of  $\Delta P_g$  against  $W_g$  or  $\Delta P_a$  against  $W_a$ , since actual isothermal measurements are at slightly different fluid rates.

$$\Delta P = \Delta P_{iso} \left(\frac{T_{av}}{T_{iso}}\right)^{1 \cdot 13} + \left(\frac{G}{3600}\right)^{2} \frac{1}{\gamma_{ig}} \left(\frac{T_{z}}{T_{1}} - 1\right)$$

$$\Delta P' = \Delta P \times \frac{12}{62 \cdot 3} \text{ in. } H_{2}O$$
(14)

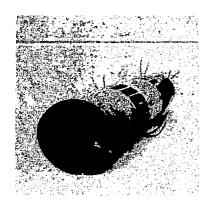


Figure 1.— Photograph of corrugated-flute type heater (stainless steel).

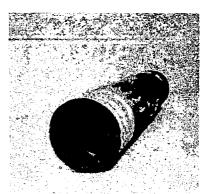


Figure 2.- Photograph of corrugated-flute type heater (copper).



Figure 3.- Photograph of air shroud used on flute-type heaters.





Figures 4 and 5 - Photograph of heaters in test stand.

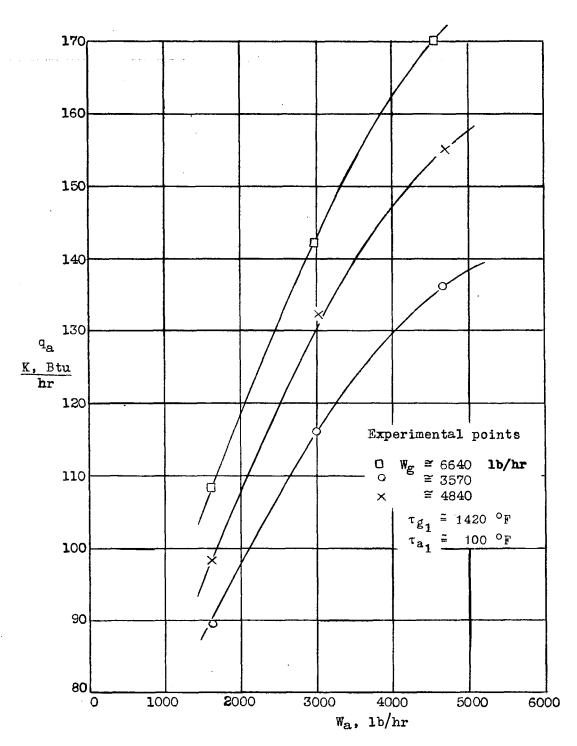


Figure 6.- Thermal output of non-corrugated flute type heater as a function of ventilating-air rate.

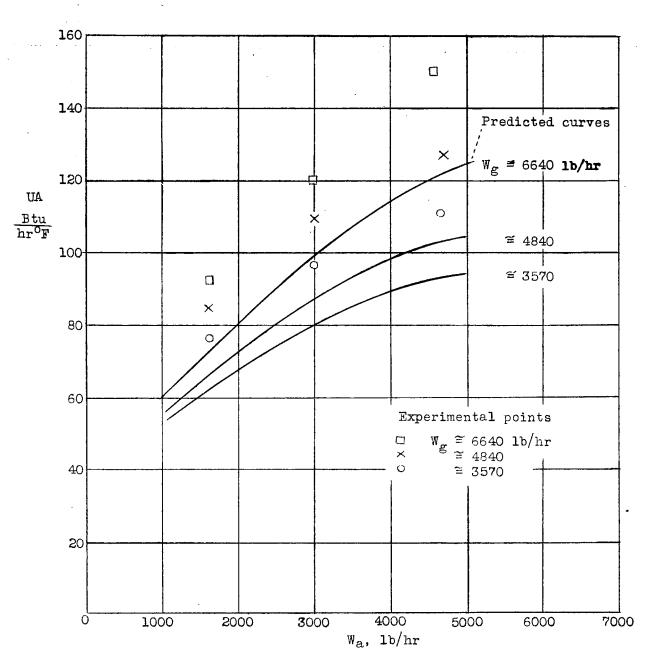


Figure 7. - Over-all thermal conductance of noncorrugated flute type heater as a function of ventilating-air rate.

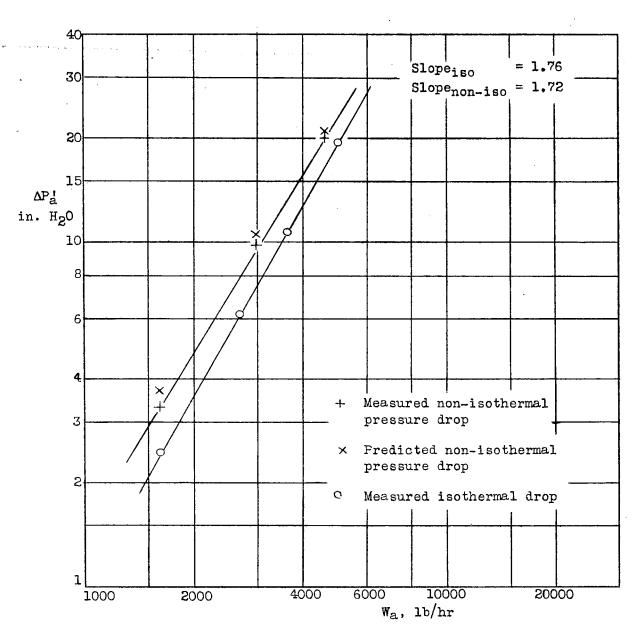


Figure 8.- Static pressure drop on air side of noncorrugated flute type heater as a function of ventilating-air rate.

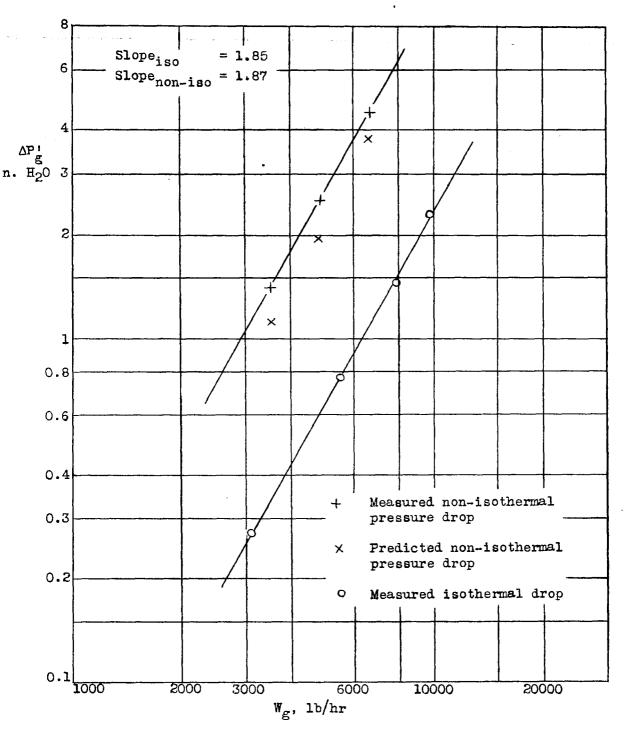


Figure 9.- Static pressure drop on the gas side of a non-corrugated flute type heater as a function of the exhaustgas rate.

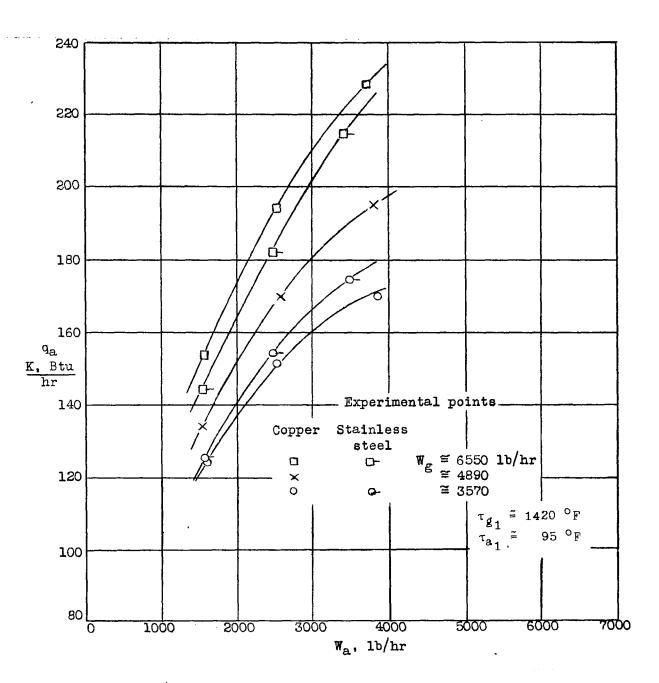


Figure 10.- Thermal output of corrugated-flute type heater as a function of ventilating-airrate.

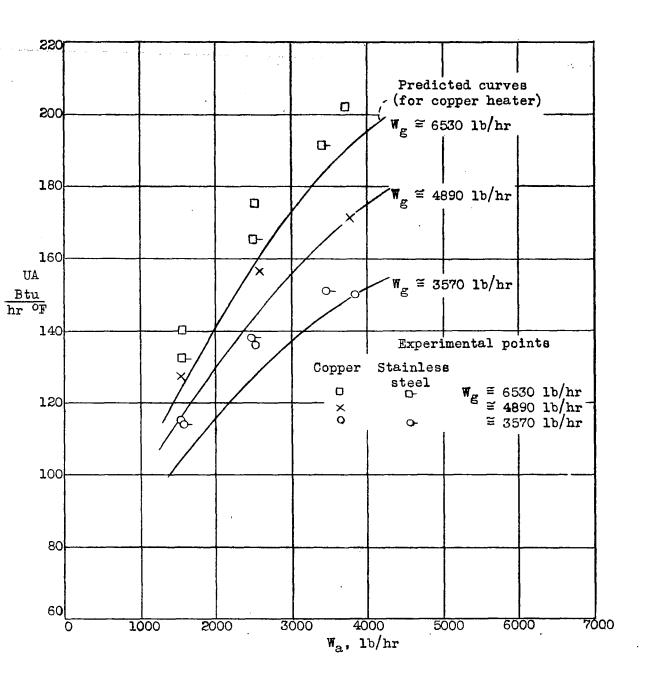


Figure 11.- Over-all conductance of corrugatedflute type heater as a function of ventilating-air rate.

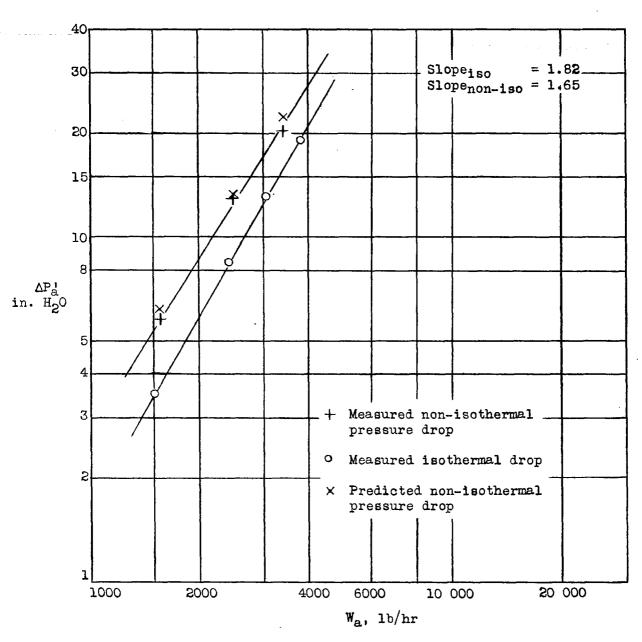


Figure 12.- Static pressure drop on ventilating-air side of a corrugated-flute type heater (stainless steel) as a function of ventilating-air rate.

1

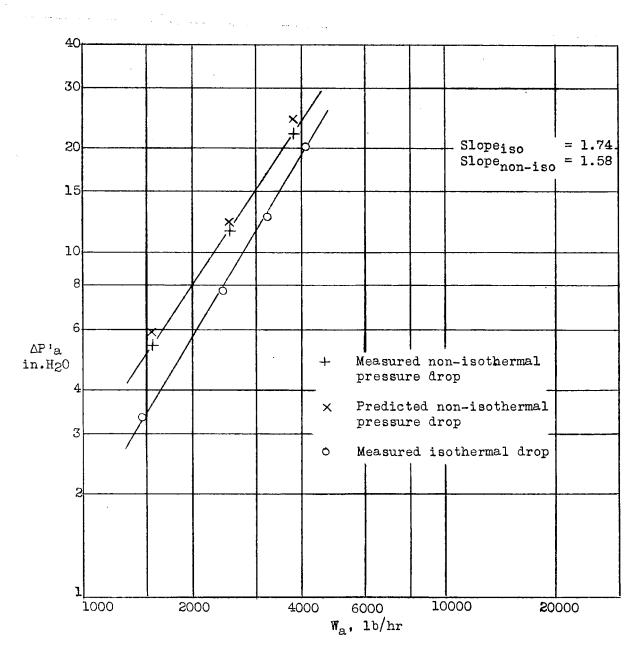


Figure 13.- Static pressure drop on ventilating-air side of corrugated-flute type (copper) heater as a function of ventilating-air rate.

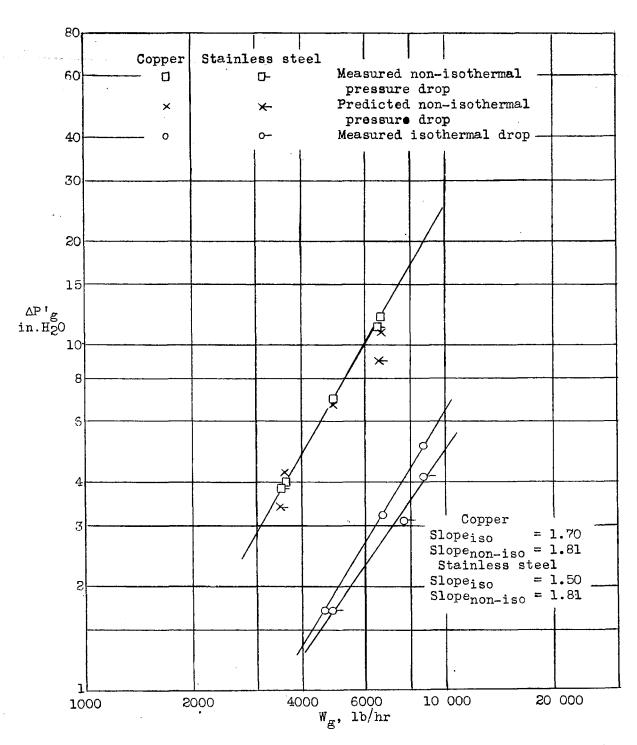


Figure 14.- Static pressure drop on exhaust-gas side of corrugated-flute type heater (copper and stainless steel) as a function of exhaust-gas rate.

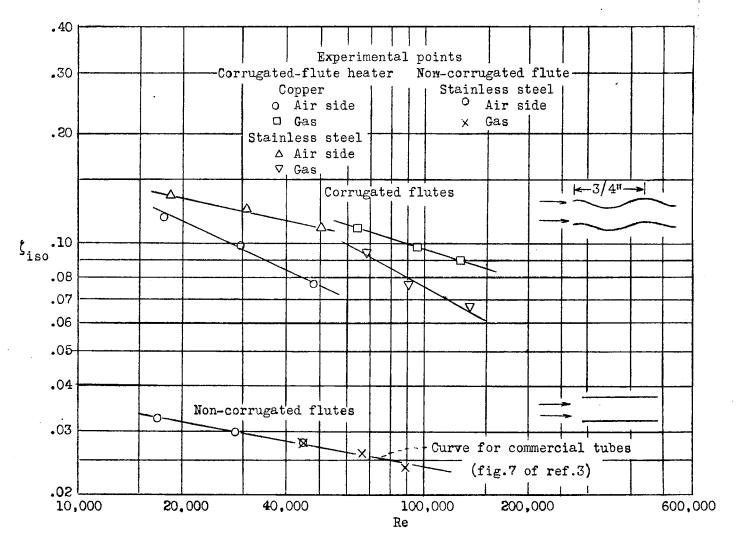


Figure 15.- Isothermal friction factor as a function of Reynolds number.

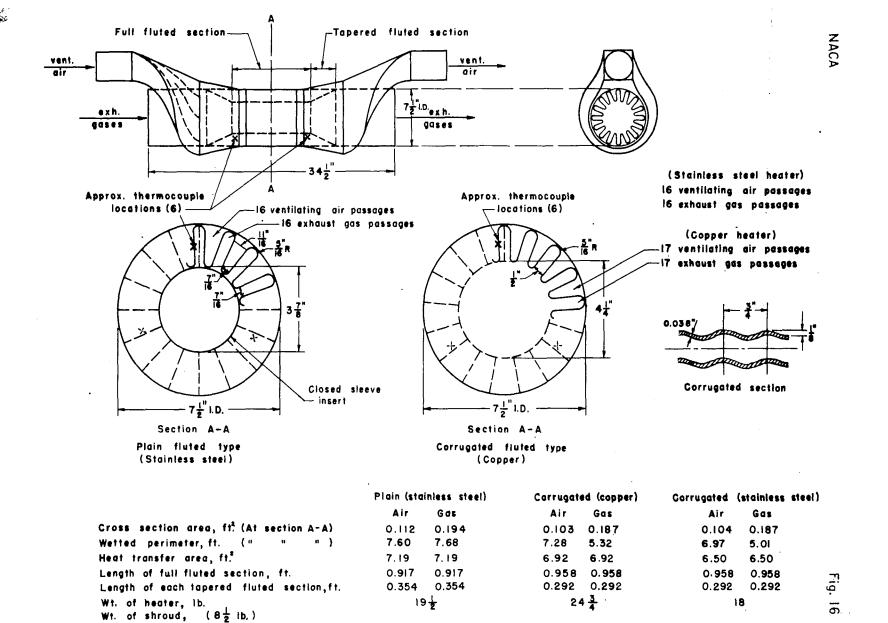


Fig. 16. - SCHEMATIC DIAGRAM OF SOLAR HEATERS AND AIR SHROUD.

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